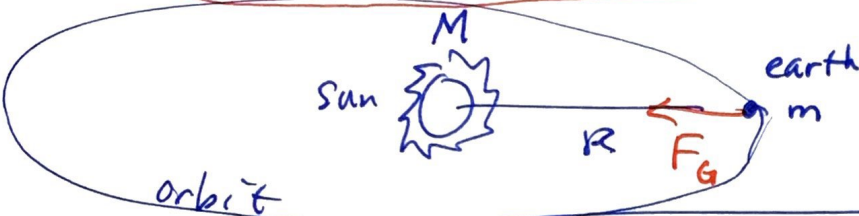


SB

Gravitational Force

(1)

Newton surmised that there is a force between the Moon and the Earth since the moon is not traveling in a straight line.



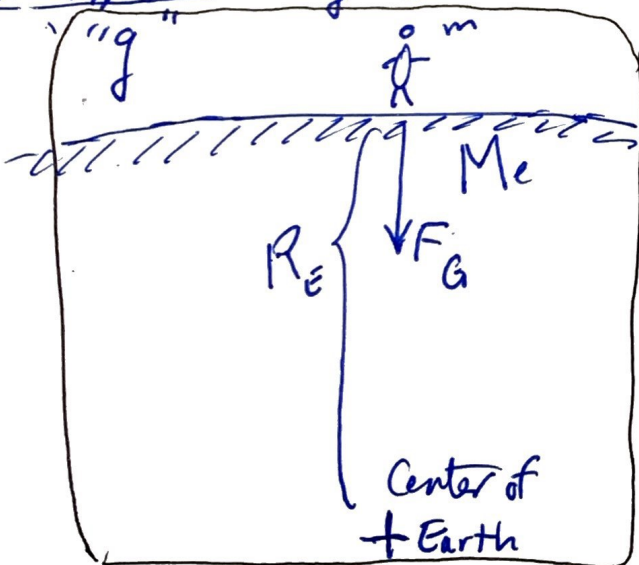
$$F_G \propto M, m, \frac{1}{R^2}$$

• Add const. of proportions...

$$F_G = G \frac{Mm}{R^2}$$

"Fourth Law"
 $G =$ gravitational constant
 $G = 6.67 \times 10^{-11} \frac{Nm^2}{kg^2}$

Q: What about little "g"?



$$F_G = G \frac{M_E m}{R_E^2}$$

$$F_G = \left(G \frac{M_E}{R_E^2} \right) m$$

$$F_G = g m$$

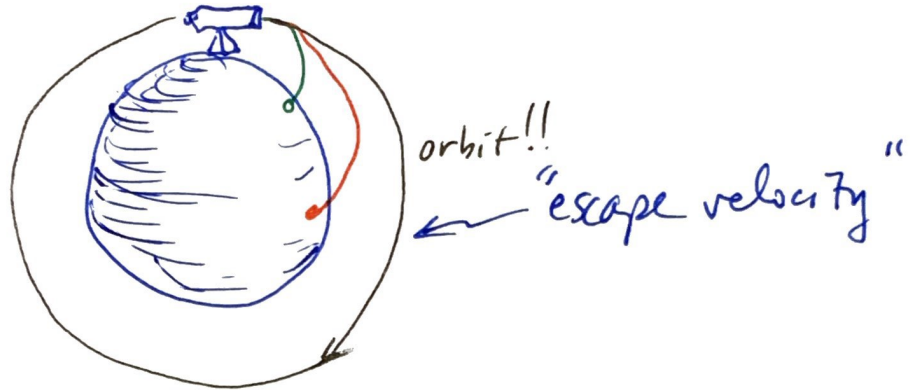
$$g = G \frac{M_E}{R_E^2} = \left(6.67 \times 10^{-11} \frac{Nm^2}{kg^2} \right) \frac{(5.98 \times 10^{24} kg)}{(6.38 \times 10^6 m)^2}$$

$$= 9.799 m/s^2$$

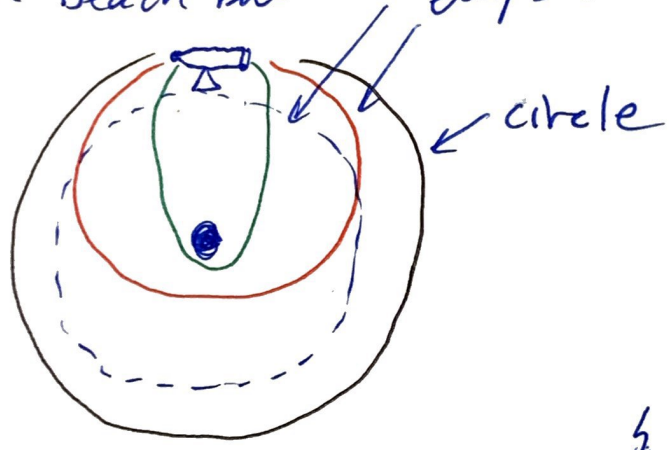
So $g_{Earth} = 9.8 m/s^2$

⊗ Orbits

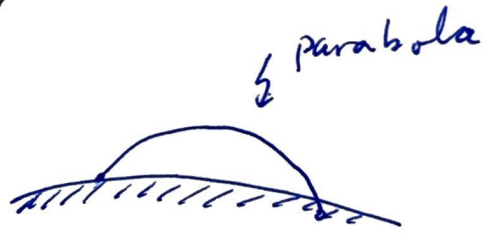
Newton surmised that if a cannonball is shot horizontally out of a cannon it could orbit the earth



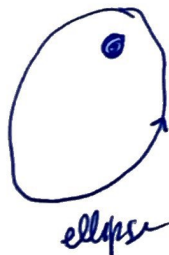
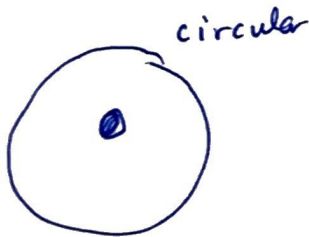
- Shrink earth beach ball



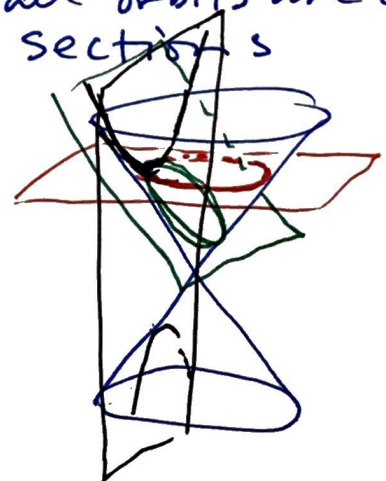
- local, near surface, the piece of the ellipse looks like a parabola.



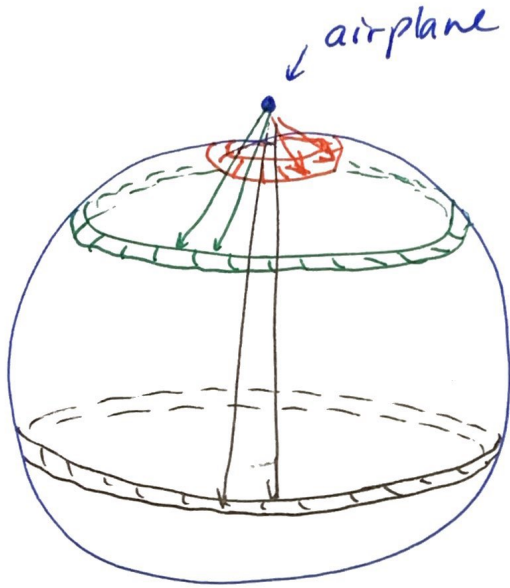
- other orbits exist



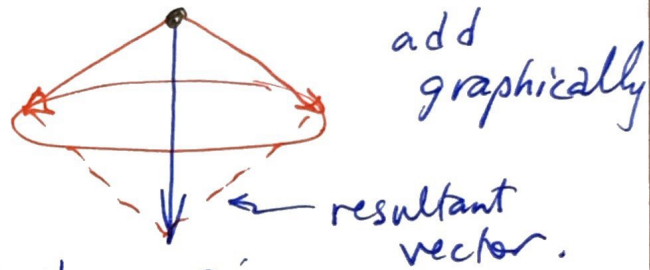
- all orbits are conic sections



* Spherical Objects (Earth)

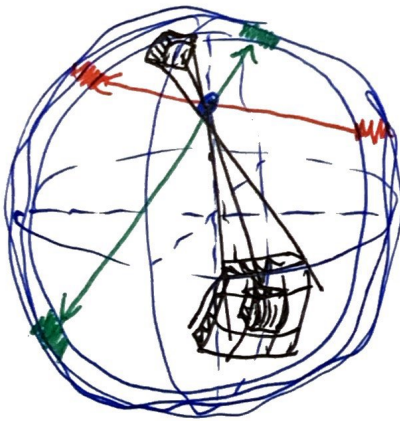


- EVERY piece of earth attracts the plane towards it.
- Due to symmetry the forces add to one pointing towards the center.



- In calculus we can add all these pieces and sum them up in a triple integral {see Wikipedia: "Shell Theorem" for the math}

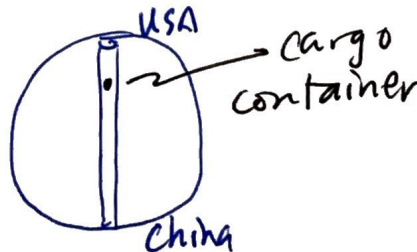
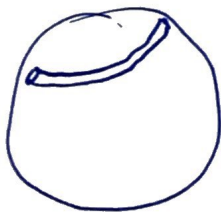
- Inside of a spherical shell of mass (Hollow Earth)



Due to the inverse square law, $\frac{1}{R^2}$, for each piece of mass on the shell, there is R^2 times mass on the opposite side of the shell that cancels out

→ an astronaut floats in a hollow planet

- Tubes in a solid Earth would provide "free" transportation



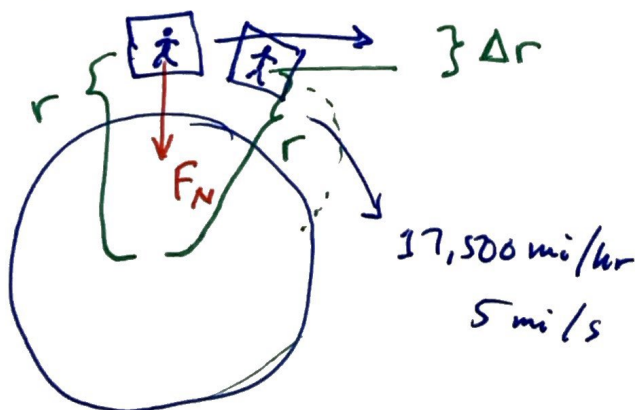
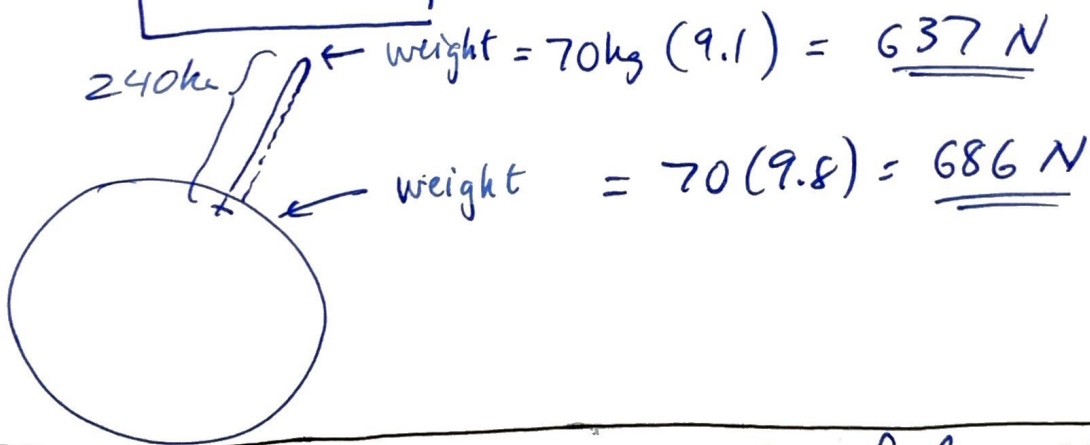
Drop in hole, it acc'ds to center then slow to a stop at the other end. Grab it!!

EX the ISS is at 240 km above the surface of the Earth. What is "g" up on station? (4)

$$g_{ISS} = G \frac{M_E}{(R_E + h)^2}$$

$$= \frac{(6.67 \times 10^{-11} \frac{Nm^2}{kg^2}) (5.98 \times 10^{24} kg)}{(6.38 \times 10^6 m + 240,000 m)^2}$$

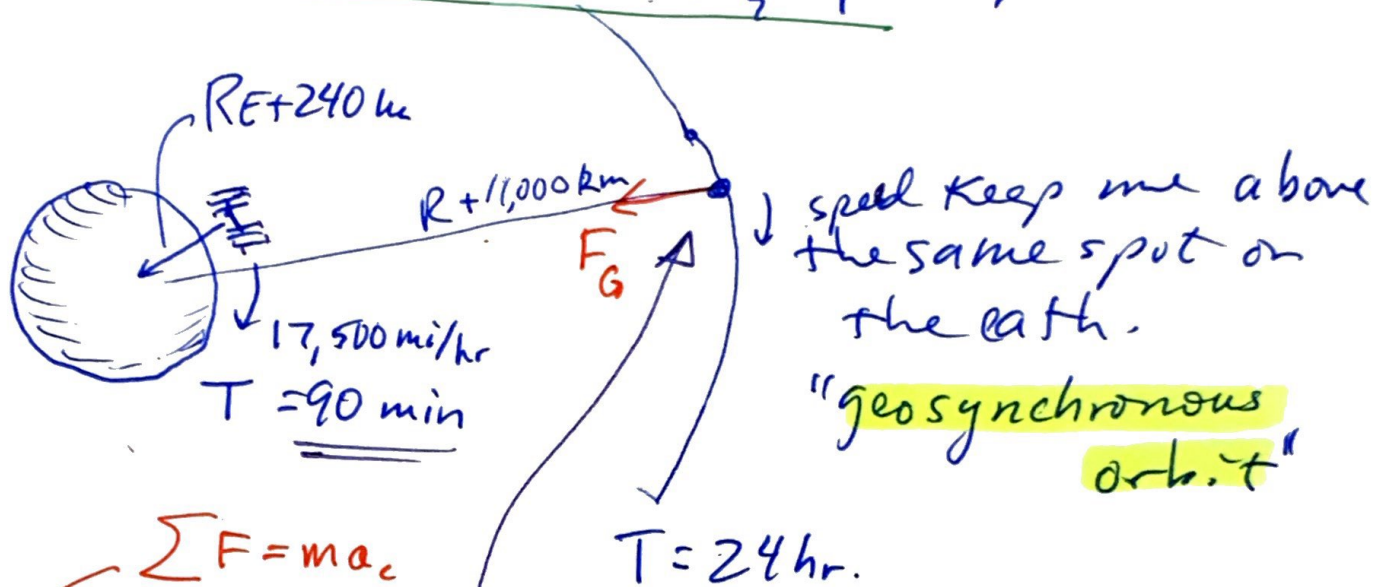
$$= \boxed{9.10 \text{ m/s}^2}$$



You feel weightlessness because you have motion parallel to surface, even though you fall towards the surface. That parallel speed keeps you from missing the Earth.

The further away we travel from Earth, the slower we need to travel to maintain constant orbital distance. ⑤

Q: what is this relation between $v \propto \frac{1}{r}$?



$$\sum F = ma_c$$

$$F_G = m_{\text{sat}} a_c$$

$$G \frac{M_E \cancel{m_{\text{sat}}}}{(R_E + h_{\text{sat}})^2} = \cancel{m_{\text{sat}}} \frac{v^2}{(R_E + h_{\text{sat}})}$$

• let $R = R_E + h_{sat}$ { center of Earth - to - satellite }

$$G \frac{M_E}{R^2} = \frac{V^2}{R}$$

$$V = \sqrt{G M_E / R}$$

EX At the ISS, how fast does the station need to be traveling to keep orbit constant?

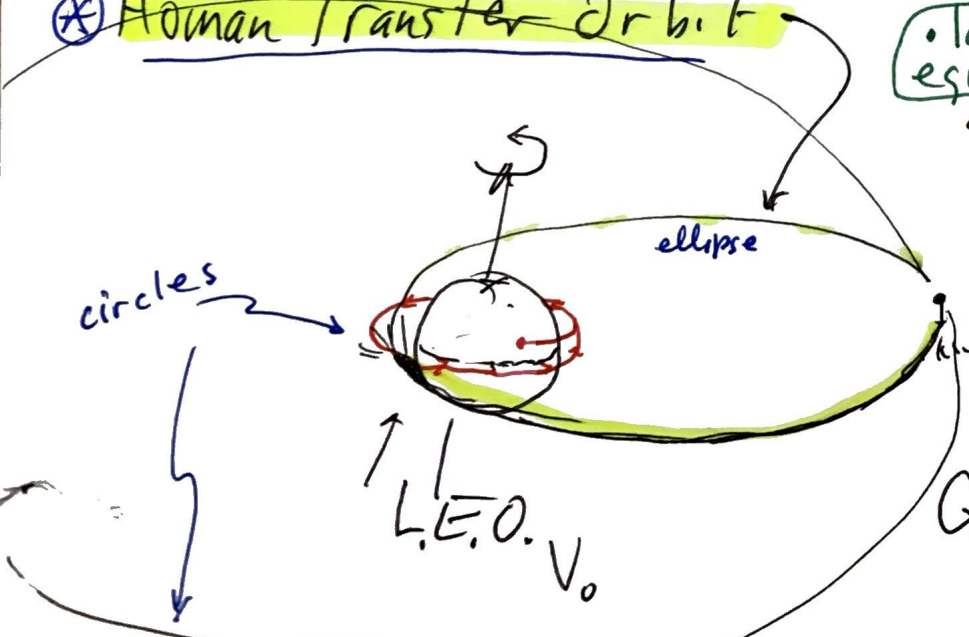
$$V = \sqrt{\left(6.67 \times 10^{-11} \frac{m^2 N}{kg^2}\right) \left(5.98 \times 10^{24} kg\right) / \left(6.38 \times 10^6 + 2.4 \times 10^5\right)^2 m}$$

$$= \boxed{7,762 m/s} \left(\frac{1 km}{1000 m}\right) \left(\frac{3600 s}{1 hr}\right) N = km/s$$

$$= 27,943.9 km/hr = \boxed{17,363.6 mi/hr}$$

17,500 mph

Homan Transfer Orbit



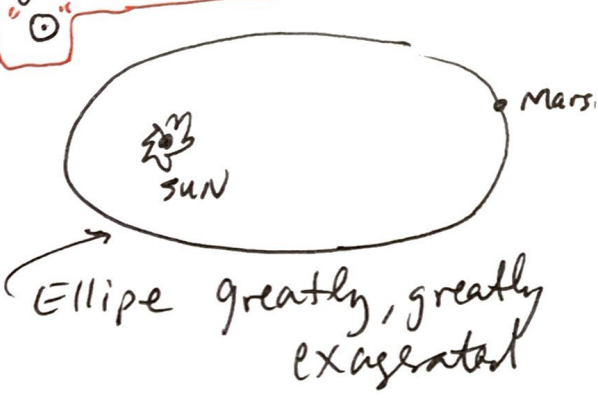
• launch to the East near equator to get help from Earth's rot'n
 • **aerodynamics**
 play "Kerbal Space" game
 • "Delta Vee" used to

$$\Delta V = V_f - V_0$$

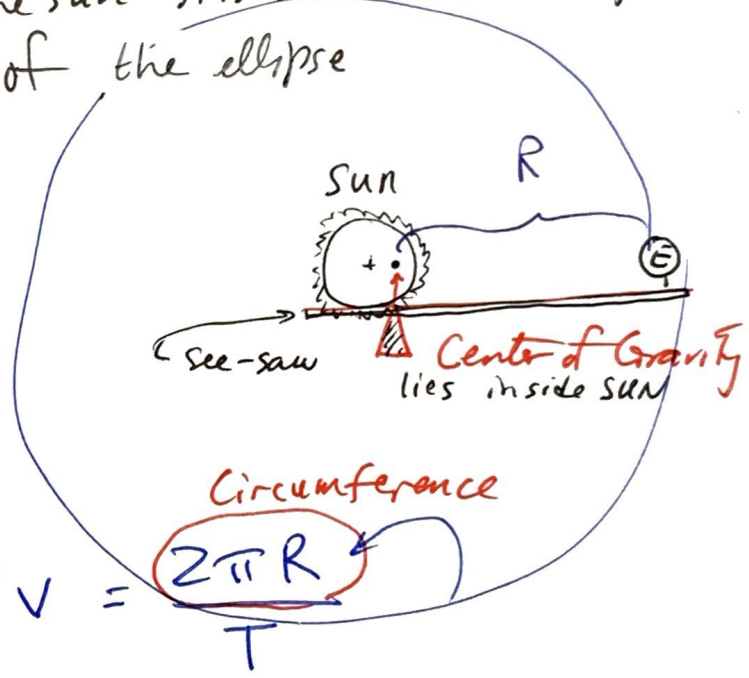
describe the added velocity to get to larger orbits

* Kepler's Law

Symbol used for the SUN



- Orbits are not perfect circles, but ellipses. (7)
- $R_{orbit}^3 \propto T_{period}^2$
- The sun sits at one focus "point" of the ellipse



• What is Kepler's Law?

$$F_G = m \frac{v^2}{R}$$

$$G \frac{Mm}{R^2} = \frac{v^2}{R} \quad \text{but} \quad v = \frac{2\pi R}{T}$$

$$G \frac{M}{R^2} = \frac{4\pi^2 R^2}{R \cdot T^2} \rightarrow \left(\frac{GM}{4\pi^2} \right) T^2 = R^3$$

$$T^2 \propto R^3$$

- Newton's 4th Law: Analytically showed what Kepler did numerically.
- Compare orbits and periods...

$$\frac{T_1^2}{R_1^3} = \text{constant} = \frac{T_2^2}{R_2^3}$$

↑ planet 1

↑ planet 2

Ex what is the distance from the center of the Earth to Geosynchronous Orbit {GSO}?

Kepler's Law :

$$R = \sqrt[3]{G \frac{M_e T^2}{4\pi^2}}$$

period of orbit of GSO satellite

$\sqrt[3]{a} = a^{1/3}$

$$R = \left[\frac{(6.67 \times 10^{-11} \frac{Nm^2}{kg^2}) (5.98 \times 10^{24} kg) \left[24 \text{ hrs} \cdot \frac{3600s}{hr} \right]^2}{4\pi^2} \right]^{1/3}$$

$$R = 42,250,174.31 \text{ m}$$

$$= 42.23 \times 10^6 \text{ m}$$

$$= \boxed{4.23 \times 10^4 \text{ km}}$$

42,300 km from center of Earth.

* Gravity is one of the 4 fundamental forces of Nature. The other 3 are

- Electromagnetic force { Physics 07 }
- Weak-force between sub atomic particles
- Strong-Force - force between protons & neutrons

